# Parameter Deviation Effect Study of the Power Generation Unit on a Doubly-Fed Induction Machine-based Shipboard Propulsion System

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 $v_f$ ,  $v_{f0}$ 

Abstract— To lower the difficulty of fault protection, a doubly-fed induction machine based shipboard propulsion system (DFIM-SPS) that is partially power decoupled is presented. In such an intrinsically safe SPS architecture, a synchronous generator (SG) is employed for power generation, and the accuracy of the parameters of power generation unit (PGU) plays an important role in SPS stable operation. In this paper, the PGU parameter deviations are studied to evaluate the effects on system performance. The models of salient-pole SG, type DC1A excitation system (EXS) and DFIM are illustrated first. Besides, the corresponding control scheme is explained. For the 16 important parameters of PGU, up to 40% of parameter deviations are applied to implement parameter sensitivity analysis. Then, simulation studies are carried out to evaluate the parameter deviation effects on system performance in detail. By defining three parameter deviation effect indicators (PDEIs), the effects on the PGU output variables, which are the terminal voltage and output active power, are studied. Moreover, the increasing rates of PDEIs with different degrees of parameter deviations for the key parameters are analyzed. Furthermore, the overall system performance is investigated for the two most influential PGU parameters. This paper provides some vital clues on SG and EXS parameter identification for DFIM-SPS.

Index Terms—shipboard propulsion system, partially power decoupled, doubly-fed induction machine, synchronous generator, parameter deviation

# NOMENCLATURE

V, I, E	Constant values of voltage, current, internal voltage
$v, i, \varphi, f$	Instantaneous values of voltage, current, flux, frequency
P, Q, PF	Active power, reactive power, power factor
$L_{ls}, L_{lr}$	Stator and rotor leakage inductance
$L_{ss}, L_{s}, L_{r}$	Inductances on the source side converter, stator, rotor $(L_s = L_m + L_{ls}; L_r = L_m + L_{lr})$
S, u	Switching signal, control signal

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X, Z	Reactance, impedance
R, L, C	Resistance, inductance, capacitance
$\theta_{\varphi},\theta_{e},\theta_{m},\theta_{slip}$	Flux angle, synchronous angle, rotor angle, slip angle
$\omega_e,\omega_m,\omega_r$	Synchronous angular frequency, mechanical rotor angular speed, electrical rotor angular speed
$T_{em}, T_L$	Electromagnetic torque and load torque
$n_p, H, F$	Number of pole pairs, inertia constant, friction factor
p	Differential calculator d/dt
$v_f, v_k$	Field and damping voltages
$V_{stab}$	Stablizing voltage
$K_a, K_e, K_f, K_d$	Regulator, exciter, stabilizer gains, damping factor
$k_s, k_p, k_i$	Stator coupling factor, proportional gain, integral gain
$X_m, X_l, X_k$	Mutual leakage and damping reactance
$T_d, T_{do}, T_{kd}$	d-axis short-circuit, open-circuit and damping time constants
$T_q, T_{qo}$	q-axis short-circuit and open-circuit time constants
$T_a, T_e, T_f, T_r$	Time constants for regulator, exciter, stabilizer, and low-pass filter
$R_{mtr}$ , $L_{mtr}$	Transformer magnetization resistance and inductance
PWM, LPF	Pulse width modulation, low-pass filter
<u>Subscripts</u>	
s, r, ss, t	Stator-side, rotor-side, source-side, total variables
0, o	Initial and open-circuit variables
n	Nominal values
f, k, T	Field winding, damper winding, terminal variables
ref	Reference value
a, b, c	Phases A, B, C
com, SG	Variables for compensator and sychronous generator
dc	DC-bus variables
$\alpha, \beta/d, q$	Direct and quadrature components referred to the stationary/synchrnous reference frame
<u>Superscripts</u>	
,	Transient variables
,,	Sub-transient variables

Field voltage, initial field voltage

# I. INTRODUCTION

\*Note: The variables defined for DFIM are not specified.

NOWADAYS, power electronic interfaces are extensively applied for all-electric ships to enhance the controllability of shipboard power system (SPS) [1]. Therefore, DC SPSs develop fast in recent years [2-4]. Nevertheless, the dominant status of power electronic devices in DC power transmission based SPS threatens the system reliability due to the easy-to-breakdown feature [5], and high difficulty of DC power system fault protection is unavoidable [6-8]. Besides, in the case of large-scale power conversion modules, usually the DC-bus voltage can be high that presents potential safety hazards [9]-[12].

Since doubly-fed induction machines (DFIMs) are capable of variable-speed constant-frequency operation, bidirectional power flows and small power converter volume, they have been widely used in wind turbines [13]. DFIM functions as the bridge between squirrel cage induction machines [14] and synchronous machines [15]. Such a kind of multiport electrical machine is endowed with high power rating, high fault tolerance, and strong power flow control capability [16], making it attractive for SPS applications [17, 18].

A DFIM-based power system without power electronic converter on the stator side is partially power decoupled. There are two paths for power flows, namely the direct and power electronic paths for the stator and rotor of DFIM, respectively. Only the slip power is controlled by using a back-to-back power converter (BTBPC) to minimize the volume of on-board integrated power electronics. However, compared with the application of DFIM in wind energy conversion systems, a much wider range of rotor speed is required for a propulsion system. In this case, a number of research activities that modify the conventional DFIM-based power system structure for SPS application were carried out [19]-[24].

In order to achieve the full speed range, the commonly adopted method is applying an additional power inverter for the stator side of DFIM [19]-[22]. Although reliable operation of DFIM propulsion systems is achieved by the aforementioned solutions with a full speed range, the intrinsic strength of a DFIM-based power system of small converter volume is lost. To retrieve the advantage of controlling the SPS with reduced power electronics and obtaining the full speed range simultaneously, the concept of "switched doubly fed machine" (switched DFM) was proposed in [23]. The design process was discussed for a switched DFM drive in [23], then the control architecture was investigated in [24] to realize the seamless transition between the two operation modes to minimize the volume of power electronics.

The power supply onboard a DFIM-SPS is mainly provided by the power generation unit (PGU) composed of a synchronous generator (SG) and its excitation system (EXS). Because of aging of power components and high temperature, the variations in SG parameters may occur, which can seriously deteriorate the system performance. In addition, the parameter deviations of the EXS also influence the overall system performance. According to the authors' knowledge, none of the previous research on DFIM considered the PGU based on SG for propulsion system power supply, not to mention the effects of parameter deviations of PGU on system performance.

The fundamental concepts of synchronous machine (SM) stability were investigated in detail by taking the effects of excitation control into account in [25], which provided guidance for the research in SM-based power systems. In addition, based on the study of modelling process without simplifications, a method that determines the SM parameters was proposed in [26]. On top of that, to ease the computational burden, some techniques with necessary simplifications were developed for the sake of exploring SM-based power system transients [27]-[29]. In order to provide satisfactory voltage control performance for an SG when there are uncertainties in

the system operating conditions [30] and exciter parameters [31], several design approaches for EXS were proposed. In [32], online calibration of sensorless position estimation was implemented by considering parametric uncertainties to improve the accuracy of position estimation. Besides, the SG and EXS parameters can be identified concurrently online by a multistage genetic algorithm-based measurement scheme proposed in [33]. Furthermore, with the purpose of overcoming the difficulty in conventional parameter identification method when the saturation effect is presented, based on d-q axes tests, both the accuracy and precision of SG parameter estimation are improved with the saturation effect taken into consideration [34]. Nevertheless, the single-machine infinite bus system is usually utilized for verifying various methods, which is not fully adaptable for a DFIM-SPS. The effects of parameter deviations of PGU on the performance of DFIM-SPS needs to be further investigated as the partially power decoupled system architecture cannot be equivalent to a general case of infinite grid.

This paper first explains the detailed models of SG and its EXS, and then up to 40% of parameter deviations are presented in the 16 key PGU parameters for a DFIM-SPS in the high-speed operation mode. The DFIM is modelled by regarding the two input electrical ports as a synthetic internal voltage [35, 36], and the control effects of phase-locked loop (PLL) fade away by using a novel vector control method. The parameter deviation effects for 16 PGU parameters on the terminal voltage and active power output are extensively studied with respect to the variations in three parameter deviation effect indicators (PDEIs) when different degrees of parameter deviations are applied. Finally, after the two most influential PGU parameters are identified, the performance of DFIM-SPS by applying the proposed novel control method is evaluated for the cases with different degrees of parameter deviations for these two parameters.

The parameter deviation effects in the PGU of DFIM-SPS are extensively investigated from different aspects by defining three PDEIs, which provides a promising way of evaluating the system performance. In addition, the most influential PGU parameters on affecting the system performance are identified, which is of paramount importance for properly designing the key system parameters to avoid unstable system operation.

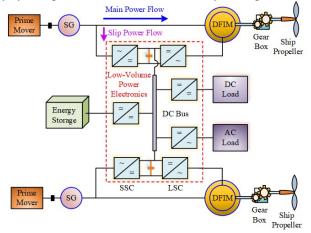


Fig. 1. The system structure of DFIM-SPS.

The paper is organized in the following structure: In Section II, DFIM-SPS modelling is illustrated. In Section III, the novel vector control strategy is described. Then, three PDEIs are developed to analyze the DFIM-SPS performance affected by parameter deviations of PGU parameters. In Section V, simulation results are displayed. Finally, Section VI presents the conclusion.

# II. MODELLING OF DFIM-SPS

The basic power components in a set of DFIM-SPS contain an internal combustion engine, an SG with EXS, and a DFIM with BTB power converter connected between its stator and rotor sides. The BTBPC includes the source side converter (SSC) and load side converter (LSC), and they are employed to control the DC-bus voltage and operation of DFIM, respectively. Fig. 1 shows the system structure of DFIM-SPS.

# A. Standard Synchronous Generator

In this paper, a PGU consisting of a salient-pole SG and a type DC1A EXS functions as the power supplier for DFIM-SPS. The stator, field and damper windings of a salient-pole SG are taken into consideration, and the rotor parameters are equivalent to the stator side. The d-axis equivalent circuit contains both the field and damper windings, while there is no field winding in the q-axis one. The salient-pole SG dq equivalent circuit model is shown in Fig. 2.

The dq electrical model of a standard salient-pole SG can be expressed by the equations as shown below.

$$\begin{cases} \vec{v}_{dq} = RsG\vec{i}_{dq} + p\vec{\psi}_{dq} + j\omega sG\vec{\psi}_{dq} \\ v_{fd} = R_{fd}\vec{i}_{fd} + p\psi_{fd} \\ \vec{v}_{kdq} = R_{kd}\vec{i}_{kdq} + p\vec{\psi}_{kdq} \end{cases}$$
(1)

$$\begin{cases} \psi_{d} = L_{d}i_{d} + L_{md}(i_{fd} + i_{kd}) \\ \psi_{q} = L_{q}i_{q} + L_{mq}i_{kq} \\ \psi_{fd} = L_{fd}i_{fd} + L_{md}(i_{d} + i_{kd}) \\ \psi_{kd} = L_{kd}i_{kd} + L_{md}(i_{d} + i_{fd}) \\ \psi_{kg} = L_{kq}i_{kg} + L_{mq}i_{q} \end{cases}$$

$$(2)$$

On the other hand, the mechanical behaviour is explained according to the following equations.

$$\Delta\omega_{SG}(t) = \frac{1}{2H} \int_{0}^{t} (T_m - T_{em}) dt - K_d \Delta\omega_{SG}(t)$$
 (3)

$$\omega_{SG}(t) = \Delta \omega_{SG}(t) + \omega_{SG0} \tag{4}$$

The reactance values of SG in the dq reference frame are calculated as shown below.

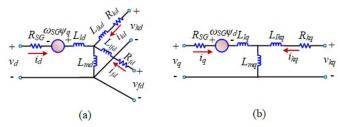


Fig. 2. SG electrical model in dq frame (a) d-axis (b) q-axis

$$\begin{cases} X_{d} = X_{ld} + X_{md} \\ X_{d}' = X_{ld} + (X_{md} || X_{lf}) \\ X_{d}'' = X_{ld} + (X_{md} || X_{lf} || X_{lkd}) \\ X_{q} = X_{lq} + X_{mq} \\ X_{q}'' = X_{lq} + (X_{mq} || X_{lkq}) \end{cases}$$
(5)

# B. EXS

The EXS is made up of an exciter, a terminal voltage transducer and load compensator, excitation control elements, and a power system stabilizer and the supplementary discontinuous excitation controls [37].

The inputs for the excitation control elements include the reference, stabilizing and compensating voltages, and the over and under excitation limiters are presented to implement saturations. The exciter is employed to produce the field voltage for the SG. Meanwhile, the voltage sensing and compensation are realized by using the load compensator and terminal voltage transducer, respectively. Additionally, the stabilizing function can be obtained by power system stabilizer and supplementary discontinuous excitation controls.

# C. DFIM

DFIM is a type of induction machine with wound rotor windings, and the electromechanical energy conversion is realized by energizing both the stator and rotor circuits. With such a configuration, there is a wide speed range for the SPS when the ship runs at a high speed, and the strong overload capability is presented. The dq electrical model of DFIM is shown as follows.

$$\begin{cases} \vec{\mathbf{v}}_{sdq} = R_s \vec{\mathbf{i}}_{sdq} + p \vec{\boldsymbol{\varphi}}_{sdq} + \mathbf{j} \omega e \vec{\boldsymbol{\psi}}_{sdq} \\ \vec{\mathbf{v}}_{rda} = R_r \vec{\mathbf{i}}_{rda} + p \vec{\boldsymbol{\varphi}}_{rda} + \mathbf{j} (\omega_e - \omega_r) \vec{\boldsymbol{\psi}}_{rda} \end{cases}$$
(6)

$$\begin{cases} \vec{v}_{rdq} = Rr\vec{i}_{rdq} + p\vec{\varphi}_{rdq} + j(\omega_e - \omega_r)\vec{\psi}_{rdq} \\ \vec{\psi}_{sdq} = L_{ls}\vec{i}_{sdq} + L_m(\vec{i}_{sdq} + \vec{i}_{rdq}) \\ \vec{\psi}_{rdq} = L_{lr}\vec{i}_{rdq} + L_m(\vec{i}_{sdq} + \vec{i}_{rdq}) \end{cases}$$
(7)

The expressions of the electromagnetic torque  $T_{em}$  and the mechanical rotor speed  $\omega_m$  are

$$T_{em} = 1.5 n_p L_m (i_{rd} i_{sq} - i_{rq} i_{sd})$$
 (8)

$$\omega_m(t) = \frac{1}{2H} \int_0^t (T_{em} - T_L) dt + \omega_{m0}$$
 (9)

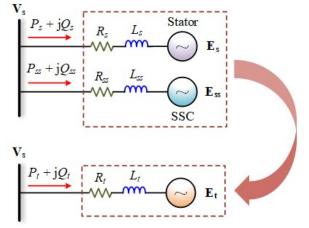


Fig. 3. The equivalent circuit of the two electrical input ports for DFIM-SPS

DFIM is an induction machine with two electrical input ports and one mechanical output port. The two input ports can be characterized as two internal voltages for the stator and SSC terminals,  $E_s$  and  $E_{ss}$ , respectively [35, 36]. For the convenience of analyzing the system, the equivalent internal voltage for the whole DFIM-SPS can be derived, whose equivalent impedance is calculated by

$$R_t + j\omega_e L_t = \frac{1}{1/(R_s + j\omega_e L_s) + 1/(R_{ss} + j\omega_e L_{ss})}$$
(10)

The stator and SSC side electrical input ports are presented for DFIM-SPS in Fig. 3. According to Fig. 3, and by substituting (10) into (9), the *dq* components of the internal stator voltage of DFIM is expressed as

$$\vec{E}_{sdq} = \vec{v}_{sdq} - R_s \vec{i}_{sdq} - pL_s \vec{i}_{sdq} + j\omega_e L_s \vec{i}_{sdq}$$
 (11)

Comparing (11) with (6) and (7), the expressions of  $E_{sd}$  and  $E_{sq}$  are modified as

$$\vec{E}_{sdq} = j\omega_e L_m \vec{i}_{rdq} \tag{12}$$

# D. Overall System Structure

The DFIM-SPS can then be represented by the combination of PGU and the equivalent internal voltages of the stator and SSC of DFIM, which can be replaced with the synthetic internal voltage  $E_{\rm t}$ .

$$\vec{E}_{s} = \frac{R_{ss} + j\omega_{e}L_{ss}}{(R_{s} + j\omega_{e}L_{s})(R_{ss} + j\omega_{e}L_{ss})} \vec{E}_{s} + \frac{R_{s} + j\omega_{e}L_{s}}{(R_{s} + j\omega_{e}L_{s})(R_{ss} + j\omega_{e}L_{ss})} \vec{E}_{ss}$$
(13)

Substituting (10) into (13), expression for  $E_t$  is updated as

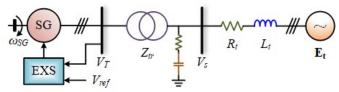


Fig. 4. The system under investigation

$$\vec{E}_{t} = \frac{R_{t} + j\omega_{e}L_{t}}{R_{s} + j\omega_{e}L_{s}} \vec{E}_{s} + \frac{R_{t} + j\omega_{e}L_{t}}{R_{ss} + j\omega_{e}L_{ss}} \vec{E}_{ss}$$
(14)

By neglecting the resistances, (14) can be simplified as

$$\vec{E}_t = L_t[(1/L_s)\vec{E}_s + (1/L_{ss})\vec{E}_{ss}]$$
 (15)

By replacing the two electrical ports with the synthetic internal voltage  $E_t$ , the system under investigation is derived as shown in Fig. 4.

# III. CONTROL OF DFIM-SPS

# A. Type DC1A EXS

As the type DC1A EXS is widely used in industry, it is taken as the one used in the proposed DFIM-SPS in this paper. The amplitude of the SG stator voltages is obtained through a low-pass filter (LPF), which is also the terminal voltage  $V_T$ . The other three inputs to the EXS are the reference voltage  $V_{ref}$ , stabilizing voltage  $V_{stab}$  and initial field voltage  $V_{f0}$ . The control signal is produced through the use of a first-order automatic voltage regulator (AVR), after which an exciter is applied to

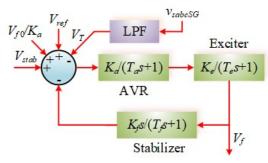


Fig. 5. Field voltage control logic by DC1A EXS

generate the field voltage  $V_f$ . To keep the system stable, a stabilizer is employed to feed the field voltage signal back. The corresponding control block diagram is displayed in Fig. 5.

# B. DFIM

For a DFIM-SPS, the PGU consisting of an SG and its EXS provides power to the propulsion load, and the PGU parameter changes can directly affect the system operation. Usually the stator resistance of DFIM is small enough to be neglected. In this case, the PLL can be eliminated to avoid possible system performance deterioration caused by improper controller parameter design, and the synchronous angle can be estimated based on the flux angle  $\theta_{\varphi}$ . Therefore, the DFIM vector control structure can be simplified, which is displayed in Fig. 6.

A steady DC-bus voltage and sinusoidal three-phase input current waveforms are to be achieved by the control of SSC. The DC-bus voltage can be calculated according to the difference between the SSC input active power  $P_{ss}$  and the LSC output active power  $P_r$ . The DC-link voltage dynamics are

$$V_{dc} = \sqrt{\frac{2(P_{ss} - P_r)}{sC_{dc}}} \tag{16}$$

$$i_{ssdref} = (V_{dcref} - V_{dc})(k_{p1} + k_{i1}/s)$$
(17)

For LSC, the electromagnetic torque is controlled by controlling the d-axis rotor current, while the stator reactive power is controlled based on the control of q-axis rotor current. The rotor dq current reference values are obtained by

$$i_{rdref} = \frac{(\omega_m - \omega_{mref})(k_{p3} + k_{i3} / s)}{k_s |\varphi_s|}$$
(18)

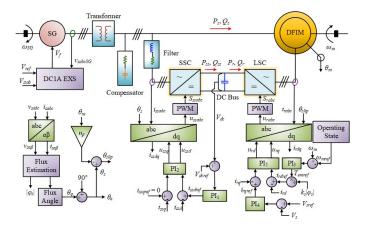


Fig. 6. Vector control of DFIM-SPS with PLL elimination

$$i_{rqref} = (V_{sref} - V_s)(k_{p4} + k_{i4}/s)$$
(19)

Where  $k_s = L_m/L_s$  is the stator coupling factor.

# IV. PARAMETER DEVIATION EFFECT INDICATORS FOR THE SYSTEM

The accuracy of PGU parameters is of paramount importance for the steady and reliable operation of DFIM-SPS. However, variations in theses parameters are inevitable in practice. In this section, 16 important parameters for SG ( $X_d$ ,  $X_d$ ',  $X_d$ ',  $X_q$ ',  $X_l$ ,  $T_{do}$ ',  $T_{do}$ '' and  $T_{qo}$ '') and its EXS ( $K_a$ ,  $T_a$ ,  $K_e$ ,  $T_e$ ,  $K_f$ ,  $T_f$ ,  $T_r$ ) are taken into account. The effects on the output terminal voltage  $V_T$  and active power  $P_e$  of SG are studied by applying different degrees of parameter deviations for each parameter. Since the input to SG is the rotor angular speed, there is no need to define the inertia and friction factor. In addition, the rotor type of the applied SG is salient-pole, under which circumstance the q-axis transient reactance  $X_q$ ' is not available. The PGU parameter values are displayed in Appendix.

In order to evaluate the parameter deviation effects on the system performance from different aspects, three parameter deviation effect indicators (PDEIs) are developed.

The average error between the output signal value (OSV) with parameter deviations and that without parameter deviation is evaluated by the average deviation (AVGD).

$$AVGD = \frac{1}{N} \sum_{i=1}^{N} (OSV_i - OSV_{iPD})$$
 (20)

where N represents the number of sampling points, i indicates the ith sampling point, and iPD indicates the value with parameter deviation at the ith sampling point.

Besides, the overall deviation of *OSV* can be measured by applying the concept of standard deviation (*STDEV*) by covering the whole sampling period, where the dispersion degree of samples is derived for each parameter.

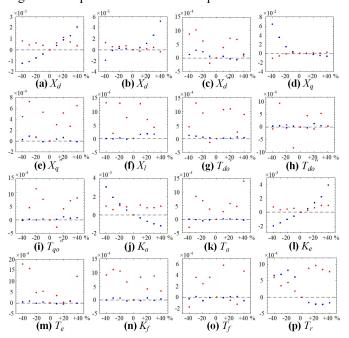


Fig. 7. AVGD for different degrees of parameter deviations of the 16 PGU parameters for  $V_T$  and  $P_e$ 

$$STDEV = \sqrt{\frac{1}{N} \sum_{i=1}^{N} \left[ (OSV_i - OSV_{iPD}) - AVGD \right]^2}$$
 (21)

Moreover, for the sake of revealing the absolute error between the values of *OSV* with and without parameter deviation, the average square root of quadratic sum (*ASRQS*) can be applied as it has a positive correlation with the absolute error.

$$ASRQS = \frac{1}{N} \sqrt{\sum_{i=1}^{N} (OSV_i - OSV_{iPD})^2}$$
 (22)

In this paper, OSV represents  $V_T$  or  $P_e$ .

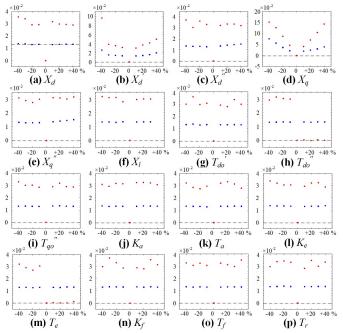


Fig. 8. STDEV for different degrees of parameter deviations of the 16 PGU parameters for  $V_T$  and  $P_\varepsilon$ 

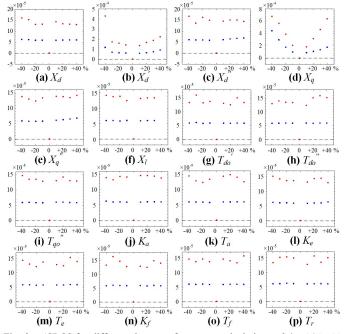


Fig. 9. ASRQS for different degrees of parameter deviations of the 16 PGU parameters for  $V_T$  and  $P_e$ 

# V. SIMULATION RESULTS AND DISCUSSION

The DFIM-SPS discrete model with the time step of  $5\mu s$  is simulated in Matlab/Simulink to investigate its operation with parameter deviations in the PGU. Since the transmission lines for the SPS are much shorter than those of a grid connected power system, the impedances of transmission lines are neglected, and the transformer impedance  $Z_{tr}$  is also set to 0. The details of system parameters are shown in Appendix.

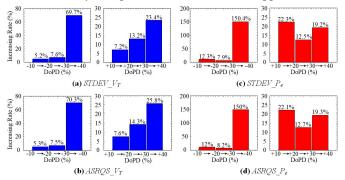


Fig. 10. PDEI increasing rates with respect to degree of parameter deviation variations of  $X_d$ 

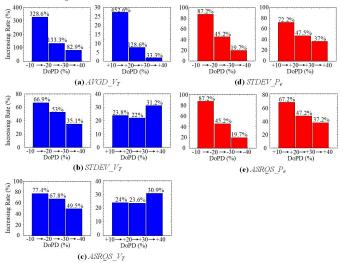


Fig. 11. PDEI increasing rates with respect to degree of parameter deviation variations of  $X_a$ 

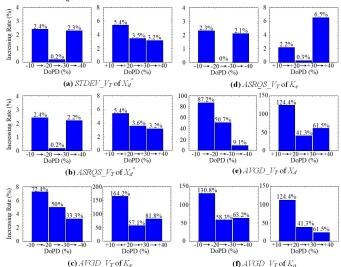


Fig. 12. PDEI increasing rates with respect to degree of parameter deviation variations of  $X_d^{"}$ ,  $K_e$ ,  $X_d$  and  $K_a$ 

With all the three aforementioned PDEIs, the parameter deviation effects can be studied roundly. The values of AVGD, STDEV and ASRQS for  $V_T$  and  $P_e$  in the cases of applying different degrees of parameter deviations for PGU are displayed in Figs. 7-9. The PDEI values for  $V_T$  and  $P_e$  are indicated by blue squares and red dots, respectively, and they are all in the unit of pu.

The degrees of parameter deviations for each parameter are set as -40%, -30%, -20%, -10%, +10%, +20%, +30% and +40% respectively from the actual value to study the parameter deviation effects on two output variables  $V_T$  and  $P_e$ . It can be seen from Figs. 7-9 that varying the degree of parameter deviations of  $X_d$  and  $X_q$  affects the values of PDEIs for  $V_T$  and  $P_e$  significantly, and the degrees of parameter deviations of  $X_a$ " and  $K_e$  majorly relate to the variations of PDEIs for  $V_T$ . In addition, the values of AVGD for  $V_T$  obviously change with the variations of degree of parameter deviations in  $X_d$  and  $K_a$ . Nevertheless, parameter deviations in the other parameters hardly influence the SG output performance. Therefore, the key parameters to be further discussed are  $X_d$ ,  $X_a$ ,  $X_d$ ,  $X_d$ ,  $X_d$ , and  $X_d$ . Moreover, the effects of parameter deviations on  $P_e$  are generally more significant than those on  $V_T$ , as can be seen from most diagrams in Figs. 7-9.

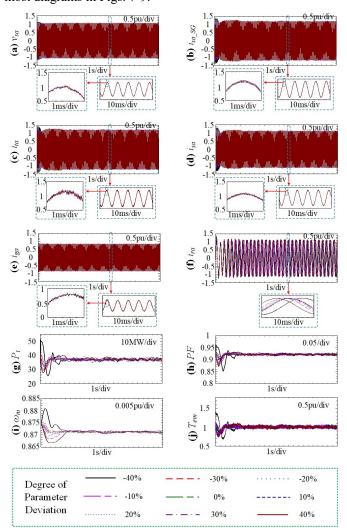


Fig. 13. Simulation results of DFIM-SPS with different degrees of parameter deviations for  $X_d$ 

The parameter deviations of  $X_d$  obviously influence *STDEV* and *ASRQS* for both  $V_T$  and  $P_e$  of SG. When applying different degrees of parameter deviations for  $X_q$ , AVGD for  $V_T$  is also significantly affected. In addition, the parameters  $X_d$ ,  $K_e$ ,  $X_d$  and  $K_a$  only have obvious impacts on the increasing rates of some PDEIs for  $V_T$  when the degree of parameter deviation varies. By increasing the degree of parameter deviation in either the negative or positive way, the increasing rates of these PDEIs vary for different regions of degree of parameter deviation variations for each parameter. In order to analyse the phenomenon in detail, the increasing rates of PDEIs with respect to degree of parameter deviation variations of  $X_d$ ,  $X_q$ ,  $X_d$ ,  $X_e$ ,  $X_d$  and  $X_a$  are shown in Figs. 10 - 12, respectively.

According to the bar graphs in Fig. 10, with the increase of degree of parameter deviation of  $X_d$ , the increasing rates of STDEV and ASRQS for  $V_T$  keep rising, while those in the variation regions of (-20%) to (-30%) and (+20%) to (+30%) are the lowest ones for  $P_e$ . Additionally, the increasing rates in the variations regions of (-30%) to (-40%) are much higher than the others for both the cases of PDEIs for  $V_T$  and  $P_e$ . In terms of  $X_q$ , the increasing rates keep falling for all the PDEIs shown in Fig. 11 except the positive variation directions for STDEV and

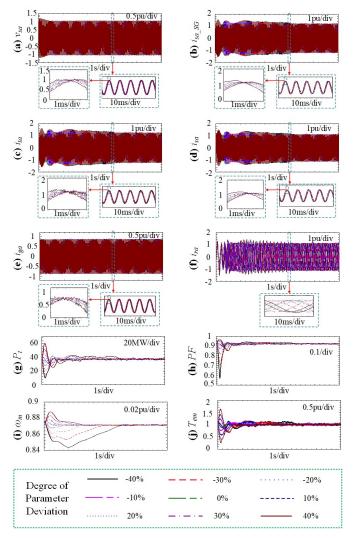


Fig. 14. Simulation results of DFIM-SPS with different degrees of parameter deviations for  $X_q$ 

ASRQS for  $V_T$ . Besides, the increasing rates of AVGD for  $V_T$  are astonishingly high for the variation regions of (-10%) to (-20%) and (+10%) to (+20%). Therefore, it can be deduced that the system performance will be significantly affected even if a slight degree of parameter deviation is applied to  $X_q$ .

From Fig. 12, it can be seen that changing the degree of parameter deviation of  $X_d$ " has little impact on the increasing rates of the corresponding PDEIs. In addition, in spite of a slightly high increasing rate (6.5%) in the region of (+30%) to (+40%) for the degree of parameter deviation of  $K_e$  in the case of ASRQS for  $V_T$ , the overall impact of degree of parameter deviation of  $K_e$  on the PDEI is negligible. The trends of increasing rate variation are the same for AVGD for  $V_T$  of  $K_e$  and  $X_d$ , and the highest values appear in the regions that are closest to the points of actual values, which is also the case for AVGD for  $V_T$  of  $K_a$ . Therefore, the deviations of  $V_T$  from the actual value will be obvious when applying a small degree of parameter deviation for  $K_e$ ,  $X_d$  and  $K_a$ .

Since the DPs of  $X_d$  and  $X_q$  play the most important role in affecting the output voltage and active power of SG, the system performance is strongly related to the degrees of parameter deviations of these two parameters. The simulation results of key system variables in the proposed DFIM-SPS are obtained when applying different degrees of parameter deviations for  $X_d$  and  $X_q$ , which are displayed in Figs. 13 and 14, respectively. The SG variables are endowed with the subscript "\_SG", while the others correspond to the DFIM variables.

In this section, the waveforms of phase-A voltage and current are displayed as representatives for the three-phase system. From Fig. 13(a), the phase-A stator voltages derived by applying different degrees of parameter deviations almost coincide with each other. However, fluctuations in the values are presented, and the larger the fluctuations are, the larger the degrees of parameter deviations are. As a result, the current waveforms also have the similar performance, which are shown in Fig. 13(b) – (e). Since the three-phase rotor currents are with a low frequency, obvious deviations of  $i_{ra}$  with 40% degree of parameter deviation (either positive or negative) from those with other values are presented. From Figs. 13(g) and (h), the oscillations of  $P_t$  and PF can be clearly observed at the beginning for the cases with large degrees of parameter deviations, while their values with different degrees of parameter deviations converge as time goes by. The situations are similar for the DFIM rotor speed  $\omega_m$  and electromagnetic torque  $T_{em}$ , which are shown in Figs. 13(i) and (j).

Different from the simulation results in Fig. 13, what is observed from Fig. 14 is that the phase-A voltage and current waveforms with different degrees of parameter deviations distinctly deviate from each other, as the degree of parameter deviation of  $X_q$  is positively correlated with AVGD for  $V_T$ . On top of that, the oscillations in the total active power  $P_t$ , power factor PF, rotor speed  $\omega_m$  and electromagnetic torque  $T_{em}$  are much larger than those in the previous case for each degree of parameter deviation. Moreover, the settling time for  $\omega_m$  and  $T_{em}$  is much longer. Therefore, the simulation results in Fig. 14 cater to the descriptions for Figs. 7-12.

# VI. CONCLUSION

The parameter deviation effects of PGU on a partially power decoupled DFIM-SPS is investigated in this paper. The modelling and control of a DFIM-SPS are studied in detail. The system performance affected by parameter deviation effects are evaluated according to three PDEIs for different degrees of parameter deviations for a specific parameter. In this paper, 16 important SG and EXS parameters are taken for sensitivity analysis to study the parameter deviation effects on terminal voltage and output active power. The overall system performance is further investigated for the two most influential PGU parameters, which are  $X_d$  and  $X_q$ . According to the simulation results, the following points are obtained to conclude the vital clues on PGU parameter identification for a DFIM-SPS.

- (1) The parameter deviations of  $X_d$  and  $X_q$  affect almost all the PDEIs for  $V_T$  and  $P_e$  significantly, except AVGD in the cases of  $X_d$  and  $P_e$  for  $X_q$ .
- (2) The parameter deviations of  $X_d$ ,  $K_e$ ,  $X_d$  and  $K_a$  have considerable influence on some of the PDEIs for  $V_T$ .
- (3) The effects of parameter deviations on  $P_e$  are generally more significant than those on  $V_T$ .
- (4) A slight degree of parameter deviation of  $X_q$  can significantly affect the system performance, as the increasing rate of AVGD for  $V_T$  when applying a small degree of parameter deviation for  $X_q$  is extremely high.
- (5) Obvious deviations of  $V_T$  from the actual value can be expected when small degrees of parameter deviations for  $K_e$ ,  $X_d$  and  $K_a$  are applied.
- (6) There is no obvious deviation for the three-phase voltages and currents when applying different degrees of parameter deviations for  $X_d$ , and  $P_t$ , PF,  $\omega_m$  and  $T_{em}$  can converge with few oscillations.
- (7) The deviations and oscillations in the corresponding variables for a specific degree of parameter deviation are much larger for the case of  $X_q$  than that of  $X_d$ , and it takes longer for  $\omega_m$  and  $T_{em}$  to be stable.

Overall, the degrees of parameter deviations of the d-axis transient reactance and q-axis reactance of SG  $X_d$  and  $X_q$  have the most obvious positive relations with the three PDEIs. Furthermore, the parameter deviations of  $X_q$  have the most significant impacts on the system performance of DFIM-SPS. Therefore, it is necessary to precisely identify the value of  $X_q$  to ensure the system stability.

#### **APPENDIX**

# A. System Parameters of DFIM-SPS

 $P_n = 36$ MW;  $V_{nSG} = 13.8$ kV;  $V_n = 4.16$ kV;  $f_n = 50$ Hz;  $C_{dc} = 10$ mF;  $R_{SG} = 0.036$ pu;  $n_{pSG} = 20$ ;  $R_s = 0.023$ pu;  $L_{ls} = 0.18$ pu;  $R_r = 0.016$ pu;  $L_{lr} = 0.16$ pu;  $L_m = 2.9$ pu; H = 3.5s;  $n_p = 3$ ;  $R_{ss} = 0.003$ pu;  $L_{ss} = 0.3$ pu; F = 0.01pu.

# B. System Parameters of PGU

 $X_d = 1.321$ pu;  $X_d' = 0.1685$ pu;  $X_d'' = 0.105$ pu;  $X_q = 1.173$ pu;  $X_q'' = 0.09$ pu;  $X_l = 0.075$ pu;  $T_{do}' = 6.5$ s;  $T_{do}'' = 0.0241$ s;  $T_{qo}'' = 0.0464$ s;  $K_a = 300$ ;  $T_a = 0.001$ s;  $K_e = 1$ ;  $T_e = 0.0001$ s;  $K_f = 0.0001$ ;  $T_f = 0.1$ s;  $T_r = 0.02$ s.

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